

## Permeability and Turbulence Over Perforated Plates

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# **Permeability and Turbulence Over Perforated Plates**

Haris Shahzad<sup>1</sup> • Stefan Hickel<sup>1</sup> • Davide Modesti<sup>1</sup>

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#### Abstract

We perform direct numerical simulations of turbulent flow at friction Reynolds number  $Re_{\tau}\approx 500-2000$  grazing over perforates plates with moderate viscous-scaled orifice diameter  $d^+\approx 40-160$  and analyse the relation between permeability and added drag. Unlike previous studies of turbulent flows over permeable surfaces, we find that the flow inside the orifices is dominated by inertial effects, and that the relevant permeability is the Forchheimer and not the Darcy one. We find evidence of a fully rough regime where the relevant length scale is the inverse of the Forchheimer coefficient, which can be regarded as the resistance experienced by the wall-normal flow. Moreover, we show that, for low porosities, the Forchheimer coefficient can be estimated with good accuracy using a simple analytical relation.

**Keywords** Wall turbulence  $\cdot$  Direct numerical simulation  $\cdot$  Perforated plates  $\cdot$  Permeable walls  $\cdot$  Forchheimer permeability

#### 1 Introduction

Turbulent flows grazing over permeable surfaces are common in engineering. Perforated plates, in particular, are used for flow conditioning (Laws and Ouazzane 1995), enhancing heat transfer in heat exchangers (Kutscher 1994), flame control in combustion chambers (Wei et al. 2017), aircraft trailing edge noise abatement (Carpio et al. 2019) and acoustic liners in aircraft engines (Casalino et al. 2018; Shur et al. 2021). Many of these applications feature turbulent grazing boundary layers over perforated plates, which result in higher drag than the baseline smooth wall. However, the drag increase is often accepted as a mandatory compromise to effectively control some other flow property, such as sound or heat transfer.

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Perforated plates are substantially different from other porous surfaces, such as metal foams, ceramic filters and gravel, for example, because they are characterized by relatively larger pores with respect to the boundary layer thickness of the grazing flow,  $d/\delta \approx \mathcal{O}(0.1)$  (Avallone et al. 2019). Pores with large diameter have the potential to substantially alter the flow physics compared to canonical porous surfaces which are characterized by large porosity (i.e. open-area ratio),  $\sigma > 0.8$ , but very small pore diameters,  $d/\delta < 0.01$  and  $d^+ = d/\delta_{\nu} < 20$  (Efstathiou and Luhar 2018; Manes et al. 2011), where  $\delta_{\nu} = v/u_{\tau}$  is the viscous length scale,  $u_{\tau} = \sqrt{\tau_w/\rho}$  is the friction velocity. As customary for flows over complex surface textures,  $\tau_w$  is the drag per plane area, which includes both the viscous and pressure drag and  $\rho$  and  $\nu$  are the fluid density and kinematic viscosity, respectively. These small pore sizes allow us to accurately model this type of surfaces using Darcy models,

$$U_{ci} = -\frac{K_{ij}}{\mu} \frac{\partial P}{\partial x_i},\tag{1}$$

where  $\partial P/\partial x_j$  is the pressure gradient across the permeable layer,  $K_{ij}$  is the permeability tensor,  $U_{ci}$  is the characteristic velocity component in i direction, and  $\mu$  is the dynamic viscosity of the fluid. Darcy permeability has the physical dimensions of an area, and it represents the ease with which flow passes through a porous surface. The Darcy law (1) stems from the momentum balance of the Navier–Stokes equations, and it is usually considered an accurate model of canonical porous surfaces, at least when the Reynolds number based on the pore diameter is small enough that the underlying Stokes approximation remains valid.

Several authors have used Darcy's boundary conditions to model the turbulent flow over porous substrates (Rosti et al. 2018; Li et al. 2020) and reported accurate results as compared to pore-resolved simulations (Kuwata and Suga 2017). With the exception of some particular configurations (Gómez-de-Segura and García-Mayoral 2019), porous surfaces tend to increase drag, similar to surface topography. (Manes et al. 2009; Manes et al. 2011) discussed the similarities and differences between canonical porous surfaces and roughness and concluded that porous surfaces interact differently with the grazing flow as compared to rough surfaces. Breugem et al. (2006), for instance, report that porous surfaces do not exhibit a fully rough regime in which the skin-friction coefficient approached a constant value with increasing Reynolds number.

The added drag provided by rough surfaces is characterized by the roughness Reynolds number  $k^+ = k/\delta_{\nu}$ , where k is the roughness height. Hence, k is usually regarded as the surface length scale to be compared with the viscous length scale,  $\delta_{\nu}$ , to determine whether the flow is strongly affected by the surface roughness. For porous surfaces, two types of length scales have generally been considered, namely the pore size d and the square root of the permeabilities  $\sqrt{K_{ij}}$ , but several authors have shown that drag depends on the dominant viscous-scaled permeability component, or a combination of  $\sqrt{K_{ij}}^+ = \sqrt{K_{ij}}/\delta_{\nu}$  (Gómezde-Segura and García-Mayoral 2019; Rosti et al. 2018; Breugem et al. 2006).

Equation (1) has been proven to be accurate for many canonical porous surfaces, and it is applicable within the limit of Stokes flow, namely for small values of the pore Reynolds number  $Re_p = \rho dU_p/\mu$ , where  $U_p$  is the velocity inside the pore. However, deviations from Darcy's law for increasing  $Re_p$  are well documented in the literature (Tanner et al. 2019; Bae and Kim 2016; Lee and Ih 2003), and have been associated with nonlinear effects that arise at high pore Reynolds numbers.  $Re_p$  is higher for perforated plates than for other porous surfaces and therefore Eq. (1) is replaced by,



$$\frac{\Delta P}{t} \frac{d^2}{\mu U_t} = \frac{d^2}{K_v} + \sigma \alpha_v dRe_p, \tag{2}$$

where  $U_t$  is the superficial velocity (Fig. 1a), t is the thickness of the plate and  $K_y$  and  $\alpha_y$  are the permeability and the Forchheimer coefficient in the direction normal to the plate, respectively (Lee and Ih 2003; Bae and Kim 2016). The superficial velocity is the hypothetical uniform incoming velocity perceived by the orifice (Tanner et al. 2019). This is related to the orifice velocity by mass conservation,

$$U_t A_p = U_p A_o \quad \Rightarrow \quad U_t = \sigma U_p, \quad \sigma = \frac{A_o}{A_p},$$
 (3)

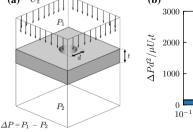
where  $A_p$  and  $A_o$  are the plate and the orifice area, respectively. Hence, the superficial velocity can be interpreted as measure of the total mass flow rate through the permeable surface.

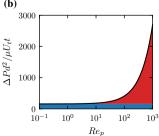
Figure 1b shows the contribution of the Darcy and Forchheimer terms to the normalized pressure drop as a function of the pore Reynolds number. It is evident that nonlinear effects start already at low values of  $Re_n \approx 5$  and become dominant at  $Re_n \approx 50$ .

At sufficiently high pore Reynolds number, Darcy drag can be assumed negligible and the entirety of the pressure drag is due to the nonlinear term. The pressure drop characteristics of perforated plates at high Reynolds numbers  $Re_p \ge \mathcal{O}(10^2)$  have been studied extensively both numerically (Tanner et al. 2019) and experimentally (Idelchik 1994; Kast et al. 2010; Malavasi et al. 2012; Miller 1990; Holt et al. 2011); however Eq. (2) for the normal flow has never been associated to the case of grazing boundary layer over porous surfaces, for which Darcy's law has always been used, to our knowledge.

In this study, we aim at clarifying the errors that potentially result from using Eq. (1) for grazing turbulent boundary layers over perforated plates, and, in particular, from using the square root of the Darcy permeability as a relevant length scale. First, we perform simulations of laminar flow through perforated plates to compute the Darcy and the Forchheimer permeability coefficients and compare the results with experimental and numerical data, and with popular engineering approximations. Second, we carry out direct numerical simulation (DNS) of turbulent channel flow grazing over perforated walls and discuss the relevance of the Forchheimer coefficient for the drag of this flow.

Fig. 1 a Sketch of the flow normal to a perforated plate with diameter d and thickness t.  $U_t$  is the superficial velocity and  $\Delta P = P_1 - P_2$  the pressure drop through the plate. b Darcy (blue) and Forchheimer (red) contributions to the pressure drop from Eq. (2) as a function of the pore Reynolds number  $Re_p$  for  $\sigma = 0.322$  and t/d = 1





## 2 Flow through a Perforated Plate

In order to calculate the Darcy and Forchheimer coefficients, we perform simulations of laminar flow through a perforated plate using the setup sketched in Fig. 1. We solve the incompressible Navier–Stokes equations, and fix the superficial velocity at the inflow and the pressure at the outflow. Neumann boundary conditions are used for the outflow velocity and inflow pressure. No-slip boundary condition is used at the surface of the perforated plate, and symmetry boundary conditions are used at the lateral boundaries.

Simulations discussed in this section are performed with the pimpleFoam solver, which is part of the open-source library OpenFOAM® (Weller et al. 1998). A forward Euler time step scheme with a maximum CFL number of 0.7 is used and simulations are run until a steady-state solution is reached (residual <  $10^{-9}$ ). The inflow and outflow boundaries are at least 40 orifice diameters away from the perforated plate. We have verified that the final solution is independent of the domain size. Approximately 10 M cells are used with a minimum mesh size of  $\approx 0.001d$  in the proximity of the plate orifice. We have performed a grid resolution study to ensure that the presented results are fully converged.

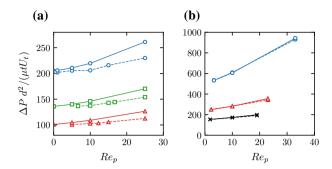
We consider 9 plate geometries with different porosity and thickness-to-diameter ratio t/d, which are summarised in Table 1. Six geometries are designed to match the parameters of Bae and Kim (2016)  $(B_{r1}-B_{r3})$ , and Tanner et al. (2019)  $(T_{r1}-T_{r3})$ , whereas  $(L_{g1}-L_{g3})$  are novel geometries. Permeability is considered independent of the spacing of the holes (Bae and Kim 2016; Malavasi et al. 2012; Tanner et al. 2019), therefore we simulate plates with a single orifice, and change the porosity by changing the orifice diameter. The pressure drop  $\Delta P$  is evaluated as the difference between the inlet and outlet pressure, see the sketch in Fig. 1a. For each of the 9 plate geometries, we perform simulations at different  $Re_p$  and use Eq. (2) to compute the Darcy permeability and the Forchheimer coefficient.

Figure 2 shows the pressure drop of flow cases  $B_{r1} - B_{r3}(\mathbf{a})$  and  $T_{r1} - T_{r3}(\mathbf{b})$  for our simulations and corresponding data from Bae and Kim (2016) and Tanner et al. (2019). We note a disagreement for flow cases  $B_{r1} - B_{r3}$  when compared to the data of Bae and Kim (2016), which becomes more evident for increasing pore Reynolds number, with

**Table 1** Forchheimer coefficient for different porosities  $\sigma$  and thickness-to-diameter ratio t/d. The last column refers to the present dataset, whereas the other columns refer to the values obtained with engineering correlations reported in the Appendix 1

	$\sigma$	t/ d	$\alpha_{_{\mathrm{y}}} d$								
			Kast et al. (2010)	Idelchik (1994)	Malavasi et al. (2012)	Miller (1990)	Holt et al. (2011)	Bae and Kim (2016)	Present		
$B_r 1$	0.2	2	12.3	6.59	12.5	6.55	4.82	7.5	7.68		
$B_r2$	0.3	2	4.69	2.36	4.62	2.35	1.75	2.91	4.67		
$B_r3$	0.4	2	2.25	1.05	2.07	1.02	0.791	1.41	2.61		
$T_r 1$	0.2	0.25	98.0	94.9	100	247	89.9	60.0	70.2		
$T_r 2$	0.4	0.25	18.0	15.2	16.6	38.5	13.4	11.2	13.17		
$T_r3$	0.6	0.25	5.55	3.69	3.91	8.28	3.69	3.33	4.43		
$L_g 1$	0.0357	1	960	643	1005	644	604	567	763		
$L_g 2$	0.143	1	51.9	33.1	53.8	33.6	25.0	31.4	46.2		
$L_g 3$	0.322	1	7.89	4.51	7.59	3.95	3.00	4.91	7.87		





**Fig. 2** Normalised pressure drop as a function of the pore Reynolds number for cases  $B_{r_1}$ - $B_{r_3}$  (**a**) and cases  $T_{r_1}$ - $T_{r_3}$  (**b**). Solid lines represent pressure drop for the current simulations. Dashed lines represent the pressure drops by Bae and Kim (2016) and Tanner et al. (2019) in (**a**) and (**b**), respectively. Symbols refer to different porosities:  $\sigma = 0.2$  (circles),  $\sigma = 0.3$  (squares),  $\sigma = 0.4$  (triangles) and  $\sigma = 0.5$  (crosses)

differences up to 20% at the highest  $Re_p$ . On the contrary, we observe a very good match for flow cases  $T_{r1} - T_{r3}$  with the data of Tanner et al. (2019). The reasons for the mismatch between the two datasets can be numerous, but are most likely down to the different flow solver and numerical set up. For example, an upwind scheme was used by Bae and Kim (2016), whereas a second-order central discretization scheme is used for the present simulations. Discrepancies of this order of magnitude are possible at high  $Re_p$  (Malavasi et al. 2012), and therefore we consider the differences acceptable.

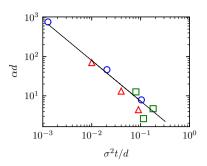
As additional validation, we compare the Forchheimer coefficients from the current simulations to several engineering correlations based on experimental data, which are summarized in Appendix 1. The values of  $\alpha_y$  returned by these correlations are reported in Table 1. There is a large spread in the Forchheimer coefficient proposed by the different correlations, and differences up to 50 - 60% seem common in the literature.

This large uncertainty of the Forchheimer coefficient can be traced back to the weak dependence of  $\alpha_y$  on  $Re_p$ , which has been reported by several studies (Tanner et al. 2019). Most of these empirical correlations are based on data at high pore Reynolds numbers in the attempt to minimize the dependence on  $Re_p$ . However, this is often not enough because the dependence of  $\alpha_y$  on  $Re_p$  can be more or less significant depending on the thickness-to-diameter ratio t/d (Tanner et al. 2019), thus complicating the evaluation of the Forchheimer coefficient. Perfect agreement with the empirical correlations is therefore not expected. The Forchheimer coefficients we calculate are within the range of the engineering correlations (50%-60% spread, Table 1), that are based on experimental data at high Reynolds number. Comparison with previous numerical simulations in Fig. 2, show a much better agreement with our data.

Even though these correlations differ from each other, they all suggest the same trend of the Forchheimer coefficient for low values of  $\sigma$ , namely  $\alpha_y \sim 1/(\sigma^2 t)$ . For this reason, we report  $\alpha_y d$  as a function of  $\sigma^2 t/d$  in Fig. 3. The figure shows a visual representation of the Forchheimer coefficient which highlights that the dependence of  $\alpha_y$  on the geometry is approximately captured with a single geometrical parameter. The approximation  $\alpha_y \sim 1/(\sigma^2 t)$  is accurate for lower values of the porosity, whereas we observe larger errors for large  $\sigma$ . For low values of the porosity  $\sigma < 0.2$  this simple analytical formula has comparable accuracy as experimental data.



**Fig. 3** Forchheimer coefficient  $\alpha_y d$  as a function of  $\sigma^2 t/d$  for  $L_{g1}$  (circles),  $B_{r1} - B_{r3}$  (squares) and  $T_{r1} - T_{r3}$  (triangles)



## 3 Turbulent flow over perforated plates

In this section, we present DNS results of turbulent grazing flow over perforated plates for different porosities and Reynolds numbers. Even though perforated plates are an elementary porous surface in terms of geometry, several configurations are in principle possible. Here, we consider geometries that resemble the acoustic liners used within aircraft engines, which consist of a perforated facesheet and a solid backplate with a honeycomb in between. Acoustic liners have an orifice diameter of about  $d/\delta \approx 0.1$  and a honeycomb depth  $h/\delta = 2$ , a porosity in the range  $\sigma = 0.05 - 0.3$ , and a plate thickness of  $t/d \approx 1$ .

## 3.1 Methodology

For the DNS, we solve the compressible Navier-Stokes equations for a perfect gas using the flow solver STREAmS (Bernardini et al. 2021). The simulations are carried out in a rectangular box of size  $L_x \times L_y \times L_z = 3\delta \times 3\delta \times 1.5\delta$ , where  $\delta$  is the channel half-width and x, y and z are the streamwise, wall-normal and spanwise directions, respectively. This box size is smaller than the one typically recommended for achieving domain independence on smooth walls (Lozano-Durán and Jiménez 2014), however, even smaller box sizes have been used successfully for studying flows over rough walls (Chung et al. 2015; Mac-Donald et al. 2017; Di Giorgio et al. 2020; Yang et al. 2022). The main idea is that when studying the flow over roughness or porous surfaces, one is interested in comparing results with a baseline smooth configuration, therefore the effect of the box size is minimized if results are compared to smooth-wall simulations with the same computational domain. The simulations are performed at bulk Mach number,  $M_b = u_b/c_w = 0.3$ , where  $u_b$  is the bulk flow velocity and  $c_w$  is the speed of sound at the wall. At this Mach number, compressibility effects are very small, and the flow can be regarded as representative of incompressible turbulence. The flow is driven in the streamwise direction by a spatially uniform body force, adjusted every time step to keep a constant bulk velocity  $u_b$ .

Periodic boundary conditions are applied in the streamwise and spanwise directions and no-slip isothermal boundary conditions are applied at the wall using a ghost-point immersed boundary method (Vanna et al. 2020). The orifice diameter is resolved with approximately 40 grid points in the streamwise and spanwise direction.

We choose the liner geometry to match the orifice size of acoustic liners in operating conditions as close as possible. The acoustic liner comprises of 64 cavities: an array of  $8 \times 4$  in the streamwise and spanwise direction on the upper and lower wall. Each cavity



**Table 2** DNS dataset comprising smooth,  $(S_n)$  and liner  $(L_n)$  cases.  $\sigma$  is the porosity (open area ratio),  $d^+$  is the orifice diameter,  $K_y$  is the Darcy permeability,  $\alpha_y$  is the Forchheimer coefficient and  $\Delta U^+$  is the Hama roughness function. Simulations are performed in computational a box with dimensions  $L_x \times L_y \times L_z = 3\delta \times 3\delta \times 1.5\delta$ 

	$Re_b$	$Re_{\tau}$	$d^+$	t/ d	σ	$\sqrt{K_y}^+$	$1/\alpha_y^+$	$\Delta U^+$	$\Delta x^+$	$\Delta y_{\min}^{+}$	$\Delta y_{\rm max}^+$	$\Delta z^+$
$S_1$	9268	506.1	0	0	0	0	0	_	5.1	0.80	3.83	5.1
$S_2$	21,180	1048	0	0	0	0	0	_	5.2	0.80	4.45	5.2
$S_3$	45,240	2060	0	0	0	0	0	-	5.2	0.80	6.67	5.2
$L_1$	9139	503.5	40.3	1	0.0357	1.04	0.0528	0.14	1.1	0.80	5.81	1.1
$L_2$	8794	496.4	39.7	1	0.142	2.06	0.859	0.56	1.0	0.80	5.81	1.0
$L_3$	8264	505.3	40.4	1	0.322	3.22	5.14	1.90	1.0	0.81	5.81	1.0
$L_4$	19,505	1038	83.0	1	0.142	4.30	1.718	0.96	2.1	0.83	6.30	2.1
$L_{u4}$	19,505	1044	83.5	1	0.142	4.32	1.727	0.98	5.9	0.84	6.10	5.9
$L_5$	17,810	1026	82.1	1	0.322	6.53	10.4	2.78	2.1	0.82	6.29	2.1
$L_6$	35,470	2044	164.0	1	0.322	13.0	20.8	4.44	4.1	0.82	6.70	4.1

 $\Delta x^+$  and  $\Delta z^+$  are the viscous-scaled mesh spacing in the streamwise and spanwise direction.  $\Delta y^+_{\min}$  and  $\Delta y^+_{\max}$  are the minimum and the maximum mesh spacing in the wall normal direction. Liner cases  $(L_1)$ – $(L_6)$  have equispaced orifices in the streamwise and spanwise direction. Case  $L_{u4}$  has the same porosity, and orifice size of  $L_4$  but the holes are not equispaced in the streamwise direction, Fig. 5

has a square cross-section with a side length  $\lambda = 0.375\delta$ , the orifices have a diameter of  $d = 0.08\delta$ , the cavity walls have a thickness of 0.5d, and the facesheet has a thickness of d. The cavities have a depth  $h = 0.5\delta$ , which is smaller than the one of typical acoustic liners. However, the cavity depth only plays a role for tuning sound attenuation and not for the aerodynamic drag Howerton and Jones (2015).

We carry out simulations at three friction Reynolds numbers in the range  $Re_{\tau} = \delta/\delta_{\nu} \approx 500 - 2000$ , corresponding to a viscous-scaled diameter of  $d^{+} \approx 40 - 160$ . Additionally, we increase the liner porosity between  $\sigma = 0.0357 - 0.322$  by varying the number of orifices per cavity between 1 and 9. An instantaneous flow visualisation of flow case  $L_{6}$  is shown in Fig. 4. Figure 4 shows the arrangements of the cavities on the upper and lower walls and also visualises the vortical structures with the aid of the Q-criterion.

**Fig. 4** Instantaneous flow field for flow case  $L_6$  at  $Re_{\tau} = 2000$  and bulk Mach number  $M_b = 0.3$ . Vortical structures are visualised using the Q-Criterion

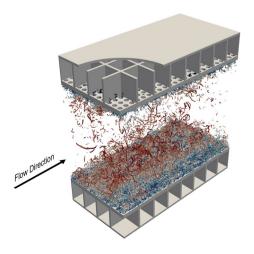
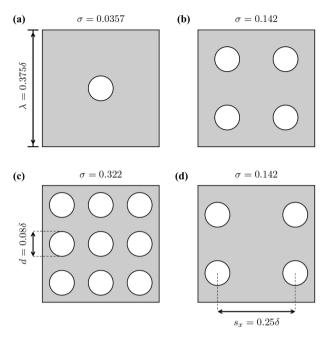


Fig. 5 Sketch the holes arrangement in the wall-parallel plane for the DNS cases. Panels (a), (b) and (c) correspond to cases  $L_{r1}$ ,  $L_{r2}$  and  $L_{r3}$ , respectively. Geometries (a), (b) and (c) have equispaced holes. Geometry (d) has the same orifice diameter, plate thickness and porosity as (b), but the orifices are not evenly spaced



The geometries considered are visualised in Fig. 5 and the complete list of flow cases is reported in Table 2. We also change the spacing between the orifices, flow cases  $L_4$  and  $L_{u4}$  (Figs. 5b, d). We compare the results of the liner simulations with smooth-wall simulations at approximately matching friction Reynolds numbers. Quantities normalized in viscous units are denoted with the '+' superscript, where  $\delta_{\nu}$  and  $u_{\tau}$  are based on the drag per plane area of the corresponding flow case.

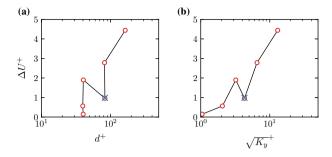
#### 3.2 Added drag and permeability

As customary for turbulent flows over rough and porous surfaces, we quantify the added drag with respect to the smooth wall using the viscous-scaled velocity deficit in the logarithmic region  $\Delta U^+$ , also referred to as Hama roughness function (Chung et al. 2021). For our geometries, we consider three candidate Reynolds numbers for scaling  $\Delta U^+$ , namely based on the orifice diameter  $d^+$ , based on the square root of the wall-normal permeability  $\sqrt{K_y^+}$  and based on the inverse of the Forchheimer coefficient  $1/\alpha_y^+$ .

Figure 6a shows  $\Delta U^+$  as a function of the viscous-scaled diameter. It is clear that  $d^+$  alone is not a suitable similarity parameter because increasing the surface porosity for a constant viscous-scaled orifice diameter leads to higher  $\Delta U^+$ . For instance, cases  $L_2$  and  $L_3$  have approximately matching  $d^+$ , but case  $L_3$  exhibits a larger  $\Delta U^+$  owing to the higher porosity.

Figure 6b shows  $\Delta U^+$  as a function of the viscous-scaled wall-normal Darcy permeability. The Darcy coefficient is also not suitable for predicting the drag increase as it does not show a consistent monotonic trend. Instead, we find that the inverse of the viscous-scaled Forchheimer coefficient  $1/\alpha_y^+$  scales very well the effect of the liner, as shown in Fig. 7a. We clearly see that  $1/\alpha_y^+$  is a promising length scale for characterising the additional drag.





**Fig. 6**  $\Delta U^+$  as a function of the viscous-scaled orifice diameter,  $d^+$  (a) and the Darcy permeability (b). Equispaced orifice cases are represented by circles. Non-equispaced orifices case is represented by a cross

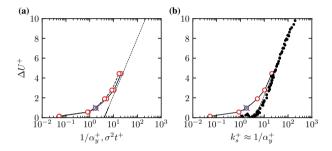


Fig. 7  $\Delta U^+$  as a function of the inverse of the Forchheimer coefficient,  $1/\alpha_y^+$  (solid line) and  $\sigma^2 t$  (dashed line) (a) and comparison with the sandgrain roughness data of Nikuradse (Nikuradse et al. 1933, filled circles) (b). The dotted line indicates the fully rough asymptote  $\kappa^{-1} \ln(1/\alpha_y^+) - 3.5$ . Equispaced orifice cases are represented by circles. Non-equispaced orifices case is represented by a cross

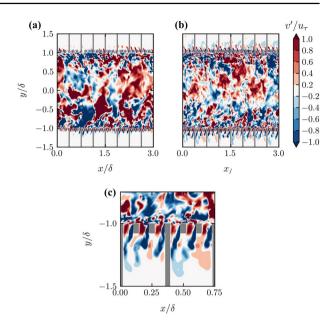
We note that the Hama roughness function tends towards the fully rough regime, and flow case  $L_6$  lies on the lower edge of the asymptote  $\Delta U^+ \approx \kappa^{-1} \ln(1/\alpha_y^+) - 3.5$ , where  $\kappa$  is the von Kármán constant. Furthermore, Table 2 shows that case  $L_{u4}$  has a nearly identical  $\Delta U^+$  to case  $L_4$ . The spacing of the orifices, therefore, has a no effect and the added drag is only a function of  $1/\alpha_y^+$ , which provides further evidence that the Forchheimer coefficient is the relevant length scale. Figure 7b shows  $\Delta U^+$  of the liner cases as a function of  $1/\alpha_y^+$ , compared to Nikuradse data of sandgrain roughness (Nikuradse et al. 1933), which suggest that acoustic liners behave like sandpaper, as  $1/\alpha_y^+ \approx k_s$ .

Additionally, we test the accuracy of the semi-empirical scaling introduced in Sect. 2,  $\alpha_y \approx 1/(\sigma^2 t)$ , and we plot  $\Delta U^+$  as a function of  $\sigma^2 t^+$  in Fig. 7a. The empirical correlation is very accurate for low values of  $\sigma^2 t^+$ , whereas minor discrepancies appear as  $\Delta U^+$  approaches the fully rough regime. This is due to the approximate correlation of the Forchheimer coefficient with  $1/(\sigma^2 t)$ , as can also be observed from the formulas in Appendix 1.

Both the Darcy permeability and the Forchheimer coefficient are suitable candidates to be considered as relevant length scales because they incorporate the effect of changes in the geometry. However, the Forchheimer coefficient clearly shows superior accuracy for the flow cases under scrutiny. This can be associated with the relevance of inertial effects inside the orifices, as we qualitatively show in Fig. 8 where we report the



Fig. 8 Wall-normal velocity fluctuations in a x-y plane at for flow case  $L_5$  (a) at  $Re_{\tau} \approx 1000$  and flow case  $L_6$  (b) at  $Re_{\tau} \approx 2000$ . Panel (c) shows a zoomed in version of (b) near the first two cavities of the lower wall to highlight the jet-like flow inside the orifices. Grey patches represent solid wall regions



instantaneous wall-normal velocity for cases  $L_5$  and  $L_6$  in a x-y plane. We observe very high wall-normal velocities inside the orifices forming a jet-like flow from the downstream edge of the orifice into the cavity, which is particularly evident for flow case  $L_6$  as fluid is pushed further inside the cavity. Inertial flow penetrating the cavities is evident in Fig. 8c, which suggests that high-wall normal velocity fluctuations below the facesheet are a strong feature characterizing this flow.

Using the maximum wall normal velocity fluctuation  $v_{rms}$  inside the orifice, we estimate a pore Reynolds number  $Re_p \approx 50-500$ , depending upon the flow case considered. The Forchheimer drag constitutes about 50% of the total drag at  $Re_p \approx 50$  and almost the entirety of the drag at  $Re_p \approx 500$ , see Fig. 1b. This is further confirmation for the use of the Forchheimer coefficient rather than the Darcy permeability as the relevant length scale for the present flow cases.

To further clarify on the relevance of the nonzero wall-normal velocity on  $\Delta U^+$  we recall that the pressure drop through the plate can be expressed in the form of friction factor in the wall-normal direction,

$$f_{y} = \frac{\Delta P}{0.5\rho U_{t}^{2}}.\tag{4}$$

In the limit of high Reynolds number, the entirety of the pressure drop can be attributed to the Forchheimer pressure drop. In such a case, Eq. (2) simplifies to

$$\frac{\Delta P}{t} \frac{d^2}{\mu U_t} = \sigma \alpha_y dRe_p, \tag{5}$$

Substitution of  $\Delta P$  from Eq. (5) into Eq. (2) leads to an expression for the Forchheimer coefficient as a function of the friction factor,



$$\alpha_{y} = \frac{f_{y}}{2t}.\tag{6}$$

Hence,  $1/\alpha_y$  represents the drag experienced by the flow normal to the plate, suggesting that  $\Delta U^+$  is intrinsically related to the wall-normal velocity fluctuations. This result is consistent with previous studies on rough surfaces that discuss the correlation between drag and wall-normal velocity fluctuations (Orlandi et al. 2006), and it reveals several similarities between roughness and porous surfaces, which have not been reported in the literature so far.

## 4 Concluding remarks

We have analysed the correlation between wall-normal permeability and wall-parallel drag in turbulent flows over perforated plates. Perforated plates are different from other types of porous surfaces because their porosity does not exceed  $\sigma \approx 0.3$  in most engineering applications, as higher values would substantially affect the structural integrity of the plate. Another main difference with respect to other porous surfaces is that the pore Reynolds number can be large, and in many applications,  $Re_p \sim \mathcal{O}(10^2)$  or higher. The result is that the Darcy equation does not hold because inertial effects inside the orifice are dominant, and the ease with which the fluid passes through the plate is better represented by the Forchheimer coefficient than by the Darcy permeability.

Accurate calculation of the Forchheimer coefficient for perforated plates is challenging, and discrepancies up to 50% are common in the literature, both from numerical and experimental sources. We calculate the Forchheimer coefficient using numerical simulations, and our results are in good agreement with a subset of the available data and engineering correlations. Semi-empirical relations for estimating the Forchheimer coefficient often show a complex dependence on the plate geometry. However, we note that in the limit of small porosity all correlations return the same functional dependence  $\alpha_y \sim 1/\sigma^2 t$ , which can be used as a first-order approximation.

In order to show the practical relevance of the Forchheimer coefficient in a realistic engineering application, we carry out direct numerical simulation of turbulent grazing flow over perforated plates, which resemble the acoustic liners used for noise attenuation over aircraft engines. We show that the inverse of the viscous-scaled Forchheimer coefficient  $1/\alpha_y^+$  is the relevant inner Reynolds number for this type of surface, and the Hama roughness function shows clear evidence of a fully rough regime. Moreover, these perforated plates provide the same drag as sandgrain roughness with  $k_s^+ \approx 1/\alpha_y^+$ . The ability of  $1/\alpha_y^+$  to represent the drag of the plate is attributed to the high values of the pore Reynolds number based on the wall-normal velocity fluctuations  $Re_p = 50 - 500$ , which suggest dominant inertial effects inside the orifice. The high r.m.s wall-normal velocity is immediately noted in the instantaneous flow visualisations.

We believe that this study sheds new light on to the interactions of a turbulent boundary layer flow with porous surfaces. We have identified the inverse of Forchheimer coefficient as a highly relevant scaling parameter, and future efforts should be directed towards an accurate numerical characterization of this length scale, both experimentally and computationally. Last but not least, we note that our findings have been verified for a considerably large data set, however, this data can cover only a fraction of the vast parameter space. We considered the effect of the porosity and orifice diameter. Their combined effect is incorporated into both the linear and non-linear permeability



of which the latter is found to be relevant for acoustic liners. Despite the relatively simple geometry, there are at least three other free parameters defining acoustic liners, namely the cavity depth, the facesheet thickness, and the cavity length and width. Future studies should certainly explore the wider parameter space of possible geometries, however, we can make some considerations based on the present results. Previous studies have addressed the effect of the cavity depth and reported no measurable change in drag (Howerton and Jones 2015), however, DNS with deeper cavities would be desirable to confirm these findings. We expect that the thickness of the facesheet could influence the added drag, and affect the relation with the sand-grain roughness height. For the geometry under scrutiny, we found  $k_s^+ \approx 1/\alpha_y^+$ , but in general one could have a proportionality relation  $k_s^+ = C/\alpha_y^+$ , similarly to what happens for different roughness geometries. As for the effect of the cavity partitions, we expect them to have a negligible effect on our conclusions, unless such modifications induce relevant streamwise and spanwise permeabilities, which can then compete with the wall-normal one for the role of relevant length scale for the flow.

# Appendix A Empirical Correlations for Pressure Drop through Perforated Plates

In this Appendix we report popular engineering formulas for estimating the Forchheimer coefficient or the friction factor.

Bae and Kim (2016) performed numerical simulations of flow through perforated plates and, proposed the following expression for the Forchheimer coefficient:

$$\alpha_{y} = \frac{3(1-\sigma)}{4\sigma^{2}t}.\tag{A1}$$

Several experimental studies at high Reynolds number are available which provide semiempirical formulas for the friction factor, which can be easily converted into Forchheimer coefficient using Eq. (6). Idelchik (1994) provides several empirical correlations for estimating the friction factor across a perforated plate. At finite thickness of the plate and high Reynolds number, Idelchik (1994) proposes a correlation of the form:

$$\alpha_{y} = \frac{1}{2\sigma^{2}t} \left( 0.5 + 0.24\sqrt{1 - \sigma}(1 - \sigma) + (1 - \sigma)^{2} \right). \tag{A2}$$

Malavasi et al. (2012) suggest an alternative relationship of the form:

$$\alpha_{y} = \frac{1}{2C^{2}\sigma^{2}t} \left(\sqrt{1 - \sigma^{2} - \sigma^{2}C^{2}} - C\sigma\right)^{2},$$
(A3)

where C is a discharge coefficient that depends upon the geometrical parameters of the orifice and the Reynolds number. Similarly, Kast et al. (2010) proposes the following relationship:

$$\alpha_{y} = \frac{1}{2\sigma^{2}t} \left( \left( \frac{1}{C} - 1 \right)^{2} + (1 - \sigma)^{2} \right).$$
 (A4)

According to Miller (1990) the Forchheimer coefficient can be expressed as:



$$\alpha_{y} = \frac{C_{0}(1 - C_{c}\sigma)}{C_{c}^{2}\sigma^{2}t},$$
(A5)

where  $C_0$  is a coefficient that depends on t/d and  $C_c$  is the jet contraction coefficient. Holt et al. (2011) present a piecewise function for the Forchheimer coefficient,

$$\alpha_{y} = \begin{cases} \frac{1}{2t} \left( 2.9 - 3.79 \frac{t}{d} \sigma^{0.2} + 1.79 \left( \frac{t}{d} \right)^{2} \sigma^{0.4} \right) K & \frac{t}{d} \sigma^{0.2} < 0.9 \\ \frac{1}{2t} \left( 0.876 - 0.069 \frac{t}{d} \sigma^{0.2} \right) K & \frac{t}{d} \sigma^{0.2} > 0.9, \end{cases}$$

where 
$$K = 1 - 2/\sigma + 2/\sigma^2(1 - 1/C_C + 1/(2C_C^2))$$
.

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#### **Declarations**

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